



Supplemental Type Certificate

Number: SVR 520

This certificate issued to: AMT Helicopters Pty Ltd
Unit 6/19 Lear Jet Drive
Caboolture, QLD 4510

certifies that the change in the type design for the following product when installed in accordance with the limitations and conditions specified herein meets the airworthiness requirements listed in the continuation sheet.

Original Product:	Type Acceptance Certificate Number:	R15, R23
	Referenced TC Number (if applicable):	FAA H2SW, TCCA H-92
	Make:	Bell Helicopter Textron
	Model:	206A, 206B
and		
	Type Certificate Number:	VR507
	Make:	AMT
	Model:	OH-58A

Description of Type Design Change: Installation of Tail Rotor Blades Part No. AMT-206-1 in accordance with Installation and Maintenance Manual CTRB-IMM, Revision IR or later approved revision.

Limitations and Conditions:

Prior to incorporating this design change, the installer shall establish that the interrelationship between this change and any other modifications incorporated into the aircraft will not adversely affect the airworthiness of the modified aircraft.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate shall remain in effect until suspended, cancelled or a termination date is otherwise established by the Civil Aviation Safety Authority.

Date of application: 17 August 2012
Date of issuance: 26 November 2014

This certificate is issued pursuant to Regulation 21.113A of the Civil Aviation Safety Regulations.



Peter Dall
Delegate of the Authority

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(Continuation Sheet)

Limitations

The following limitations are supplementary to the conditions and limitations given in the original product reference certificate. The original conditions and limitations continue to apply unless superseded by the limitations below.

Applicability

This STC is limited to Composite Tail Rotor Blades Part No. AMT-206-1 manufactured in accordance with a current CASA manufacturing approval.

Airworthiness Limitations

The Composite Tail Rotor Blade is limited to 5000 hours time-in-service (TIS).

Type Design Data

1. Master Records Index - Doc No. 209/373/MRI, Revision 5, dated 25 November 2014, or later CASA approved revision.
2. Flight Manual Supplement
 - a. For Bell 206A/B - Doc. No. 209/373/FMS1, Revision IR, dated 20 October 2014 or later CASA approved revision.
 - b. For AMT OH-58A - Doc. No. 209/373/FMS2, Revision IR, dated 20 October 2014 or later CASA approved revision.
3. Installation and Maintenance Manual – Doc No. CTRB-IMM, Revision IR, dated 18 November 2014 or later CASA approved revision.

Certification Basis for the components and areas affected:

CAR 6 effective 20 December 1956:

6.111, 6.112, 6.113, 6.114, 6.115, 6.120, 6.121, 6.122, 6.123, 6.130, 6.140, 6.225

and

FAR 27 effective 1 February 1965 including amendments up to FAR 27-44:

27.1, 27.177, 27.301, 27.303, 27.305, 27.307, 27.391, 27.395, 27.397, 27.571, 27.601, 27.602, 27.603, 27.605, 27.607, 27.609, 27.610, 27.611, 27.613, 27.619, 27.623, 27.625, 27.629, 27.653, 27.659, 27.661, 27.671, 27.1301, 27.1309, 27.1529, 27.1565, 27.1581